

UNCLASSIFIED

AD NUMBER

AD822660

LIMITATION CHANGES

TO:

Approved for public release; distribution is unlimited.

FROM:

Distribution authorized to U.S. Gov't. agencies and their contractors;  
Administrative/Operational Use; 16 NOV 1967.  
Other requests shall be referred to Space and Missile Systems Organization, Norton AFB, CA 92409.

AUTHORITY

SAMSO ltr 28 Feb 1972

THIS PAGE IS UNCLASSIFIED

AD 822660

FILE COPY

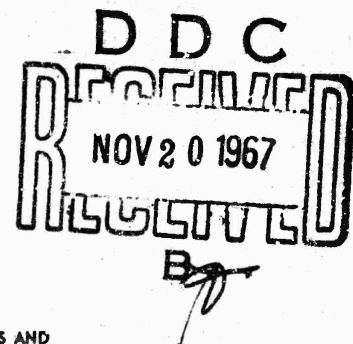
EXPERIMENTAL AND THEORETICAL KINETICS OF  
HIGH TEMPERATURE FLUOROCARBON CHEMISTRY

Prepared by

AVCO MISSILES, SPACE AND ELECTRONICS GROUP  
MISSILE AND SPACE SYSTEMS DIVISION  
AEROPHYSICS LABORATORY  
201 Lowell Street  
Wilmington, Massachusetts 01887

AVMSD-0734-67-CR  
Contract F04(694)-67-C-0060 and  
Contract DA-01-021-AMC 12005 (Z)  
CDRL Item 6

November 16, 1967



THIS DOCUMENT IS SUBJECT TO SPECIAL EXPORT CONTROLS AND  
EACH TRANSMITTAL TO FOREIGN GOVERNMENTS OR FOREIGN  
NATIONALS MAY BE MADE ONLY WITH PRIOR APPROVAL OF SPACE  
AND MISSILE SYSTEMS ORGANIZATION (SMSO).

Prepared jointly for

ADVANCED RESEARCH PROJECTS AGENCY

Monitored by the  
ARMY MISSILE COMMAND  
UNITED STATES ARMY  
Redstone Arsenal, Alabama

SPACE AND MISSILE SYSTEMS ORGANIZATION  
DEPUTY FOR REENTRY SYSTEMS  
AIR FORCE SYSTEMS COMMAND  
Norton Air Force Base, California 92409

FOR OFFICIAL USE ONLY

18 SAMS TR-67-20  
19  
6

9 Technical Rept.

This document consists of 33 pages,  
87 copies, Series A

EXPERIMENTAL AND THEORETICAL KINETICS OF  
HIGH TEMPERATURE FLUOROCARBON CHEMISTRY.

Prepared by

AVCO MISSILES, SPACE AND ELECTRONICS GROUP  
MISSILE AND SPACE SYSTEMS DIVISION  
~~AEROPHYSICS LABORATORY~~  
201 Lowell Street  
Wilmington, Massachusetts 01887

15 FO4694-67-C-0060,  
14 AVMSD-0734-67-CR  
~~FO4694-67-C-0060~~  
DA-01-021-AMC 12005(Z)  
CDRL Item 6  
11/6 Nov 1967  
12 33p.

by

10 Anthony P. Modica  
APPROVED

*G. Luceri*  
G. Luceri, Manager  
REST Program

THIS DOCUMENT IS SUBJECT TO SPECIAL EXPORT CONTROLS AND  
EACH TRANSMITTAL TO FOREIGN GOVERNMENTS OR FOREIGN  
NATIONALS MAY BE MADE ONLY WITH PRIOR APPROVAL OF SPACE  
AND MISSILE SYSTEMS ORGANIZATION (SMSO).

Prepared jointly for

ADVANCED RESEARCH PROJECTS AGENCY

Monitored by the  
ARMY MISSILE COMMAND  
UNITED STATES ARMY  
Redstone Arsenal, Alabama

SPACE AND MISSILE SYSTEMS ORGANIZATION  
DEPUTY FOR REENTRY SYSTEMS  
AIR FORCE SYSTEMS COMMAND  
Norton Air Force Base, California 92409

mt

402 572

1473  
elk

#### ABSTRACT

Ultraviolet absorption was used to measure the rate of formation of difluoromethylene ( $\text{CF}_2$ ) from decomposed  $\text{CF}_3\text{I}$ ,  $\text{C}_2\text{F}_6$ , and  $\text{CF}_4$  in excess argon behind shock waves. In some experiments pure  $\text{CF}_4$  was shocked. Data were taken over a temperature range from  $1700^\circ$  to  $3000^\circ$  K at total concentrations between  $2(10^{-5})$  and  $5(10^{-6})$  mole- $\text{cm}^{-3}$ . A chemical, nonequilibrium shock-tube computer program was developed to analyze the  $\text{CF}_2$  kinetic profiles. By curve fitting the data, rate constants for a number of fluorocarbon reactions are obtained.

Unclassified report

A copy of this report is available at Reports Distribution Center, Room 1126 and will be held for a period of three weeks.

EDITED BY:  
EDITORIAL SERVICES SECTION  
P. S. Falcey

### Foreword

This report presents work performed by the Chemical Physics Research Group of the Aerophysics Laboratory of the Avco Corporation Missile and Space Systems Division, 201 Lowell Street, Wilmington, Massachusetts. The present task has been supported jointly by the Space and Missile Systems Organization (SAMS0), Deputy for Ballistic Missile Reentry Systems, Air Force Systems Command, Norton Air Force Base, California, under Contract F04 (694)-67-0060 part of the ABRES Program; and by the Avco Everett Research Laboratory (AERL) for the Advanced Research Projects Agency (ARPA), monitored by the Army Missile Command, United States Army, Redstone Arsenal, Alabama, under Contract No. DA-01-021 AMC-12005 (Z), part of the Project DEFENDER.

Information in this report is embargoed under the Department of State to foreign governments by department or agencies of the U.S. Government subject to approval of Space and Missile Systems Organization (SMSD), Norton Air Force Base, California, or higher authority within the Department of the Air Force. Private individuals or firms require a Department of State export license.

The Air Force monitor for the ABRES REST Program is Capt. W. Mercer, SMYSE, USAF, Project Officer.

This technical report has been reviewed and is approved.

Capt. W. Mercer, SMYSE  
REST Project Office  
Space and Missile System Organization  
Norton Air Force Base, California

## CONTENTS

I.	Introduction .....	1
II.	Experimental Measurements .....	2
III.	Nonequilibrium Shock Tube Program .....	3
IV.	Chemical Mechanisms and Kinetics .....	5
V.	Discussion .....	7
VI.	References .....	9
VII.	Appendix A: Formulation of the Nonequilibrium Shock-Tube Program .....	10

## ILLUSTRATIONS

Figure 1	Difluoromethylene Ultraviolet Light Absorption Behind Shock Waves .....	18
2	Numerical Check of Nonequilibrium Shock-Tube Program with Airchemistry Stream-Tube Calculation .....	19
3	Chemical Kinetics Shock-Tube Calculation for Decomposition of 1:100 $\text{CF}_3\text{I}$ - Argon Gas Mixture .....	20
4	Chemical Kinetics Shock-Tube Calculation for Decomposition of 1:100 $\text{C}_2\text{F}_6$ - Argon Gas Mixture .....	21
5	Experimental Rate Constant for $\text{CF}_3$ Dissociation .....	22
6	Chemical Kinetics Shock-Tube Calculation for Decomposition of 1:100 $\text{CF}_4$ - Argon Gas Mixture .....	23
7	Experimental Rate Constant for $\text{CF}_4$ Dissociation .....	24
8	Chemical Kinetics Shock-Tube Calculation for Decomposition of Pure $\text{CF}_4$ .....	25
9	Kinetic Profiles of $\text{CF}_4$ in Shock Heated Fluorocarbon-Argon Gas Mixtures .....	26

## TABLES

Table 1	Fluorocarbon Reactions and Rate Constants .....	15
2	Examples of $\text{CF}_3\text{I}$ and $\text{C}_2\text{F}_6$ Shock Tube Experiments .....	16
3	Examples of $\text{CF}_4$ Shock Tube Experiments .....	17

## INTRODUCTION

The role of fluorine in the high temperature oxidation chemistry of tetrafluoroethylene ( $C_2F_4$ ) is little understood. For example, among the reaction products of  $C_2F_4$  and its polymer with oxygen, there have been identified  $CF_4$  and  $CF_3$  species.<sup>(1,2)</sup> It was suspected that these molecules resulted from the reaction between fluorine and the difluoromethylene radical ( $CF_2$ ). Support of this hypothesis was noted when shock heated mixtures of  $CF_3I$ ,  $C_2F_6$  and  $CF_4$  in argon diluent as well as pure  $CF_4$  were found to decompose to  $CF_2$ . Experiments were carried out to detect the  $CF_2$  radical by its absorption in the ultraviolet. In the present paper, measured kinetic profiles of this species are analyzed with a computer program. The computer program solves simultaneously the Hugoniot relations (mass, momentum, and energy) and the chemical rate equations for a chemically unrelaxed shock wave. The calculated rate profiles of  $CF_2$  are compared with experiment. Some of the rate constants employed in the analysis were taken from literature sources, some were determined empirically and others were estimated from theory or deduced from JANAF equilibrium constants.

### EXPERIMENTAL MEASUREMENTS

The shock-tube apparatus has been described previously.<sup>(1)</sup> It consists of 1.5-inch inside diameter stainless steel sections provided with light screens for shock velocity measurements and an optical station for ultraviolet absorption spectroscopy. Gas mixtures of  $\text{CF}_4$  (Matheson, 95 percent purity),  $\text{CF}_3\text{I}$  (Air Products, 98 percent purity),  $\text{C}_2\text{F}_6$  (Air Products, 99.9 percent) in excess argon (Matheson, 99.999 percent purity) were prepared for study. A few tests were conducted in undiluted  $\text{CF}_4$ . The  $\text{CF}_2$  radicals in the shock-heated reaction mixtures were followed photometrically in absorption at  $2660\text{\AA}$ . The absorption coefficient,  $1.15 \times 10^6$  ( $\pm 10$  percent) cubic centimeter per mole-centimeter, was used to calculate concentration from Beer's law.<sup>(3)</sup> Oscillogram records shown in Figure 1 are typical of the  $\text{CF}_2$  absorption from these gas mixtures. The  $\text{CF}_3\text{I}$  and  $\text{C}_2\text{F}_6$  molecules were studied since  $\text{CF}_3$  radicals are immediately formed behind the shock wave, making the rate determining step for  $\text{CF}_2$  appearance the dissociation of  $\text{CF}_3$ . With  $\text{CF}_4$ , the complete fluorocarbon chemistry involving  $\text{CF}_4$ ,  $\text{CF}_3$  and  $\text{CF}_2$  is given. By studying the decomposition rates leading to  $\text{CF}_2$ , the kinetics of the fluorine oxidation reactions may thus be ascertained from detail balancing with appropriate equilibrium constants.

# NONEQUILIBRIUM SHOCK TUBE PROGRAM

The equations governing a moving one-dimensional inviscid fluid in a shock tube are the so-called Rankine-Hugoniot<sup>(4)</sup> relations expressing conservation of mass, momentum, and energy,

$$d(\rho u)/dt = 0 \quad (\text{mass}) \quad (1)$$

$$\rho u d(u + \frac{p}{\rho u})/dt = 0 \quad (\text{momentum}) \quad (2)$$

$$\rho u d(h + 1/2 u^2)/dt = 0 \quad (\text{energy}) \quad (3)$$

where  $\rho$  is the density,  $u$  is the flow velocity,  $p$  is pressure and  $h$  is the enthalpy per unit mass. For a multi-component gas  $h$  is defined by

$$h = \frac{\sum c_i h_i}{\sum c_i M_i} \quad (4)$$

The quantity  $c_i$  is the concentration of the  $i^{\text{th}}$  species in mole per cubic centimeter,  $h_i$  is the molar enthalpy and  $M_i$  the gram molecular weight. The ideal gas equation

$$p = \frac{\rho \sum c_i}{\sum c_i M_i} RT \quad (5)$$

is used to relate the thermodynamic temperature  $T$  to pressure and density where  $R$  is the universal gas constant per mole. The molar enthalpy is then expressed in terms of temperature by a power series,

$$h_i = \left( L_1 T + \frac{L_2 T^2}{2} + \frac{L_3 T^3}{3} + \frac{L_4 T^4}{4} - L_5 T^{-1} + L_6 \right)_i \quad (6)$$

A curve fit of the molar enthalpy of each species was generated from JANAF thermochemical data.<sup>(5)</sup> The variation in species concentration resulting from chemical reaction and volumetric change behind the shock wave is given by

$$\begin{aligned} \frac{dc_i}{dt} - \frac{c_i}{\rho} \frac{d\rho}{dt} &= \sum \nu_{kj} c_k A_{fj} T^{n_j} \exp(E_{fj}/RT) \\ &- \sum \nu_{ij} c_i A_{rj} T^{n_j} \exp(E_{rj}/RT) \end{aligned} \quad (7)$$

where the positive term on the right of the equality reflects the sum of the chemical rates leading to species production and the negative term, the rates for removal. Subscripts  $f$  and  $r$  denote the forward and reverse processes,  $\nu_{kj}$  and  $\nu_{ij}$  are the stoichiometric coefficients for the  $k^{\text{th}}$  and  $i^{\text{th}}$  species in the  $j^{\text{th}}$  reaction;  $A_j$ ,  $n_j$  and  $E_j$  are the pre-exponential factor, temperature exponent, and activation energy of the rate constant for that reaction. To solve this set of differential equations for a multicomponent, chemically relaxing shock wave, a Runge-Kutta integration scheme<sup>(6)</sup> and an IBM 360 computer were employed. The inputs to the calculation are the shock velocity, state of the gas ahead of the shock wave, the chemical reactions to be considered, the forward and reverse rate constants, and coefficients of the enthalpy curve-fits. Typical machine times for solving the fluorocarbon chemical kinetics problems ran between 1 and 2 minutes. Before using the nonequilibrium shock-tube program to analyze the present data, a program check was made against an air chemistry, stream tube program developed by M. G. Mamin and O'Brien.<sup>(7)</sup> Figure 2 shows that calculations from both programs for electron density behind a shock wave into air agree numerically to within 3 percent.

## CHEMICAL MECHANISMS AND KINETICS

The reactions considered in this study are shown in Table 1.

CF<sub>3</sub>I and C<sub>2</sub>F<sub>6</sub> Decomposition: A bimolecular rate constant for R1 was estimated from classical collision theory<sup>(8)</sup> assuming energy contributions from half the number of vibrational modes and a steric factor of 0.1. The dissociation rate constant of R2 was obtained from the JANAF equilibrium constant and the rate constant of Ayscough<sup>(9)</sup> for trifluoromethyl radical recombination. Calculations based on these rate constants (Figures 3 and 4) showed that within a few microseconds after shock passage CF<sub>3</sub>I was dissociated by several orders of magnitude and that C<sub>2</sub>F<sub>4</sub> was about 80 percent dissociated with R2 and R2' in equilibrium. Under these conditions, the apparent rate constant for CF<sub>2</sub> formation by CF<sub>3</sub> dissociation (R3) is given by

$$\frac{d[CF_2]}{\rho_{21} dt} = k_3 (\text{expt'l}) \{ [CF_3]_0 - (1 + \gamma) [CF_2] \} [Ar] \quad (8)$$

where  $\gamma$  is zero if  $R4' \ll R3$  and one if  $R4' \gg R3$ , i.e., for each CF<sub>2</sub> produced one CF<sub>3</sub> radical is removed when fluorine attachment is slow and two CF<sub>3</sub> radicals are removed when fluorine attachment is fast. The concentration of CF<sub>3</sub> initially behind the shock wave is  $[CF_3]_0$ . The density ratio  $\rho_{21}$  is multiplied in Equation (8) to reference the event to laboratory time ( $t_l$ ). In calculating  $k_3$  (expt'l) by means of Equation (8),  $\gamma$  was taken to be unity for the lower temperature data, and was assumed zero for the higher temperature data. Calculations (Figures 3 and 4) indicated that when analyzing the low temperature data after 50 percent decomposition, the above approximation was good to within 90 percent. Another feature brought out by the calculations was that after CF<sub>3</sub>I and C<sub>2</sub>F<sub>6</sub> chemical relaxation, the shock temperature profile was nearly constant, so that little temperature uncertainty was introduced in evaluating the rate constant for R3 during reaction. With the higher temperature data,  $k_3$  (expt'l) was determined from the initial CF<sub>2</sub> absorption slope. As seen in Figure 5, the rather good agreement obtained within the scatter limits of  $k_3$  (expt'l) in the CF<sub>3</sub>I and C<sub>2</sub>F<sub>6</sub> experiments tends to uphold the data reduction method. Examples of the CF<sub>3</sub>I and C<sub>2</sub>F<sub>6</sub> study are given in Table 2.

CF<sub>4</sub> Decomposition: In the temperature range 2200° to 3000° K, the CF<sub>3</sub> dissociation rate is three orders of magnitude faster than the rate of appearance of CF<sub>2</sub> in the oscillogram records of CF<sub>4</sub> decomposition. Hence these observations suggested that the rate determining step for CF<sub>2</sub> formation starting with CF<sub>4</sub> was R4. Calculations (Figure 6) showed that under the experimental conditions chosen the CF<sub>3</sub> concentration would be much less than that of CF<sub>2</sub> and CF<sub>4</sub> in the reaction mixture, allowing an apparent rate constant for R4 to be evaluated from the initial CF<sub>2</sub> slope measurements according to

$$\frac{d[CF_2]}{\rho_{21} dt} = k_4 (\text{expt'l}) \{ [CF_4]_0 - [CF_2] \} [Ar]. \quad (9)$$

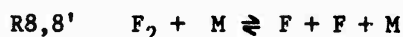
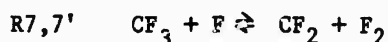
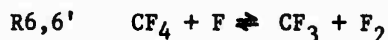
Examples of the experiments are shown in Table 3. The temperature dependence of the apparent rate constants obtained from 1:100 and 2:100  $\text{CF}_4$ -Argon gas mixtures is shown in Figure 7. The kinetic profile of  $\text{CF}_2$  measured in a pure  $\text{CF}_4$  run (Figure 8) is compared to that calculated from the nonequilibrium shocktube program using the rate constants in Table 1. The results indicate that the collision efficiencies of  $\text{CF}_4$  and argon appear to be quite similar. Another aspect of the pure  $\text{CF}_4$  experiment is that the intermediate  $\text{CF}_3$  chemistry becomes important and serves as a check on its rate constants in the calculations. The rate constants for R5 and R5' were obtained in an earlier study of the thermal decomposition of the difluoromethylene radical<sup>(10)</sup>.

Finally, for a number of shock conditions, measured and calculated kinetic profiles of  $\text{CF}_2$  from decomposition of the three fluorocarbon molecules investigated are shown in Figure 9. It is seen that the rate constants given in Table 1 would appear to be adequate in describing the experimental data over the temperature range of the study.

### DISCUSSION

The heat-of-reaction for R3 was obtained from JANAF thermochemical data and used to fit the experimental rate constants of the  $\text{CF}_3\text{I}$  and  $\text{C}_2\text{F}_6$  data. The large temperature exponent ( $T^{-9.04}$ ) determined appears to be inconsistent with the classical collision treatment for a polyatomic like  $\text{CF}_3$ . The maximum number of internal degrees of freedom not counting rotations which could contribute energy for reaction would be 6. Rotational and translational participation, if included, would increase the temperature exponent to only ( $T^{-8.5}$ ). It was found that in calculating the  $\text{CF}_2$  rate profiles, good agreement with the measurements could be obtained with the JANAF equilibrium constant multiplied by 16. With the equilibrium pre-exponential factor kept constant, a factor of 16 gives a heat-of-reaction of 81.2 kcal/mole at 2000° K (average temperature range of data). By refitting the rate constants of R3 with the revised heat-of-reaction, a temperature exponent of ( $T^{-6.21}$ ) is obtained, which is more in accord with theory.

Besides the reactions shown in Table 1, the rate data were also analyzed by adding the reactions



Calculations based on rate constants from classical collision theory and JANAF equilibrium constants showed that R6,6' and R7,7' were about three orders of magnitude slower than the decomposition reactions and had negligible effect on the measured  $\text{CF}_2$  kinetic profiles.

#### ACKNOWLEDGEMENTS

The support of the Space and Missile Systems Organization/U. S. Air Force under Contract F04(694)-67-C-0060, part of ABRES Program, is gratefully acknowledged. The authors express thanks to Ronald Brochu for assisting in the collection and reduction of the shock-tube data.

#### REFERENCES

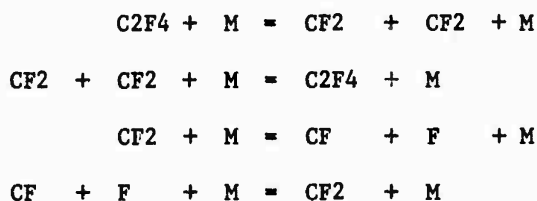
1. Modica, A. P. and J. E. LaGraff, "Decomposition and Oxidation of  $C_2F_4$  Behind Shock Waves," *J. Chem. Phys.* 43, 3383 (1965).
2. Kupel, R. E., M. Nolan, R. G. Keenan, M. Hite and L. D. Scheel, "Mass Spectrometric Identification of Decomposition Products of Polytetrafluoroethylene and Polyfluoreoethylenepropylene," *Anal. Chem.* 36, 386 (1964).
3. Modica, A. P., "Electronic Oscillator Strength of  $CF_2$ ," Avco Report AVMSD-0346-67-CR, (May 1967).
4. Courant, R. and K. O. Friedrichs, Supersonic Flow and Shock Waves (Interscience Publishers, Inc., New York, 1948).
5. JANAF Thermochemical Tables (Dow Chemical Co., Midland, Mich. 1961-1966).
6. Ralston, A. and H. S. Wilf, Mathematical Methods for Digital Computers (John Wiley and Sons, Inc., New York, 1964).
7. McMenamin, D. and M. O'Brien, "The Finite Difference Solution of Multi-component Nonequilibrium Steady Inviscid Streamtube Flows Using a Novel Stepping Technique. Part I Analysis and Applications," General Electric Co. Report 67SD241 (April 1967).
8. Fowler, R. and E. A. Guggenheim, Statistical Thermodynamics (Cambridge University Press, Cambridge, England, 1956).
9. Ayscough, P. B., "Rate of Recombination of Radicals. II. The Rate of Recombination of Trifluoromethyl Radicals," *J. Chem. Phys.* 24, 944 (1956).
10. Modica, A. P., "Kinetics and Equilibria of the Difluorocarbene Radical Decomposition Behind Shock Waves," *J. Chem. Phys.* 44, 1585 (1966).

## APPENDIX A

### Formulation of the Nonequilibrium Shock-Tube Program

The nonequilibrium shock tube program solves simultaneously the Rankine-Hugoniot conservation relations and the chemical rate equations expressing variation in species concentration. The program is organized to integrate the rate equations stepwise, an iteration scheme to satisfy the conservation relations being repeated at each step. It differs from the average ordinary differential equation program chiefly in that the equations themselves are not fixed. Instead, only rules for generating the equations are fixed.

The input quantities include the shock velocity; the temperature, pressure, and species mole fractions ahead of the shock; the chemical reactions to be considered; the preexponential factor, temperature exponent, and activation energy of the rate constant for each reaction; and coefficients of the enthalpy curve fits. A library tape of species data (molecular weights and enthalpy fit coefficients) may be searched for quantities omitted from the regular program input. The chemical reactions are entered directly in equation form, a maximum of four species allowed on each side of an equation, with each species allotted a maximum of eight alphanumeric characters. For example, consider the simple system



where M will be taken by convention to refer to the sum of all species concentrations.

The set of equations is analyzed by the program, the species in each equation matched against an input list of all species occurring in the problem, and a matrix M of indices set up to represent the original equations. If, in the example above, the species are arbitrarily ordered,

	C <sub>2</sub> F <sub>4</sub>	CF <sub>2</sub>	CF	F
1	2	3	4	

and M is assigned the index 5, the 4x8 matrix generated by the program is

$$M = \left[ \begin{array}{cccc|cccc} 1 & 5 & 0 & 0 & 2 & 2 & 5 & 0 \\ 2 & 2 & 5 & 0 & 1 & 5 & 0 & 0 \\ 2 & 5 & 0 & 0 & 3 & 4 & 5 & 0 \\ 3 & 4 & 5 & 0 & 2 & 5 & 0 & 0 \end{array} \right]$$

The rules for generating the set of differential equations are then as follows:

For the  $i^{th}$  species the change in the concentration  $C_i(t)$  as a function of time is defined as

$$\frac{d}{dt} C_i(t) = \sum_{j=1}^{N_R} R_j (\delta_{l_j} + \delta_{r_j}) \prod_{K=1}^4 C_{M_{j,K}} \quad i = 1, \dots, N_S$$

where

$N_S$  is the number of species

$N_R$  is the number of reactions

$R_j$  is the rate constant corresponding to the  $j^{th}$  equation

$$\delta_{l_j} = \begin{cases} -n & \text{if the } i^{th} \text{ species occurs } n \text{ times on the left} \\ & \text{side of equation } j \\ 0 & \text{if the } i^{th} \text{ species does not occur on the left} \\ & \text{side of equation } j \end{cases}$$

$$\delta_{r_j} = \begin{cases} +n & \text{if the } i^{th} \text{ species occurs } n \text{ times on the right} \\ & \text{side of equation } j \\ 0 & \text{if the } i^{th} \text{ species does not occur on the right} \\ & \text{side of equation } j \end{cases}$$

and by definition

$$C_0 = 1$$

$$C_{N_S+1} = \sum_{i=1}^{N_S} C_i$$

Again, for the example, the equations are

$$\frac{d}{dt} C_1 = (-R_1 C_1 + R_2 C_2^2) \sum_{i=1}^4 C_i$$

$$\frac{d}{dt} C_2 = (2 R_1 C_1 - 2 R_2 C_2^2 - R_3 C_2 + R_4 C_3 C_4) \sum_{i=1}^4 C_i$$

$$\frac{d}{dt} C_3 = (R_3 C_2 - R_4 C_3 C_4) \sum_{i=1}^4 C_i$$

$$\frac{d}{dt} C_4 = (R_3 C_2 - R_4 C_3 C_4) \sum_{i=1}^4 C_i$$

To find initial conditions for the system of differential equations, the conservation relations are first solved for values of the temperature  $T_2(0)$  and density  $\rho_2(0)$ . A false position iteration for temperature is used in the solution. Then, since the mole fractions  $X_i(t)$  are known (input) quantities at time 0, the concentrations may be found from the relation

$$C_i(t) = \frac{X_i(t) \rho_2(t)}{\sum_{i=1}^{N_S} X_i(t) M_i} \quad i = 1, \dots, N_S$$

where  $M_i$  is the molecular weight of the  $i^{th}$  species.

To solve the differential equations a fourth order predictor-corrector method, sometimes called the Modified Adams-Bashforth Method, is used. This method is generally available for computer use in an Avco library subroutine which was partially reprogrammed for purposes of the problem. The technique is particularly attractive because of its computational efficiency and the relative ease it affords in maintaining accuracy requirements. The one major disadvantage is that it is not self-starting.

Before the predictor-corrector algorithm can even be applied, values of all the concentrations  $C_i(t)$ ,  $i = 1, \dots, N_S$  and of their first derivatives  $\frac{d}{dt} C_i(t)$ ,  $i = 1, \dots, N_S$

must be known at each of four points in time. Information at the first point  $t_0 = 0$  can be obtained directly from the initial conditions. Values at three additional equally-spaced points  $t_1$ ,  $t_2$ , and  $t_3$  are obtained by a fourth order Runge-Kutta integration process, the interval  $\Delta t$  between the points being controlled by program input. Runge-Kutta is particularly appropriate as a starting method since results at each step depend only on conditions at the last previous step. In addition, a high degree of accuracy can be established with a sufficiently small interval. (However, Runge-Kutta would present serious disadvantages if used over the entire time period. A relatively large number of derivative

calculations are required, and no immediate estimate of the truncation error is available.) In the fourth order Runge-Kutta method the derivatives are evaluated at four points in the interval and weighted to give agreement with a Taylor Series expansion through the fourth order term. For a comprehensive discussion of Runge-Kutta integration see chapter 9 in Mathematical Methods for Digital Computers, edited by Ralston and Wilf(6).

Given the starting values obtained from Runge-Kutta, the predictor-corrector algorithm operates as follows. For each concentration  $C_i, i = 1, \dots, N_s$ , a third degree polynomial  $\tilde{P}_i(t)$  approximating  $d/dt C_i$  is fitted through the four available derivative values. This polynomial is then integrated to give an extrapolated or predicted concentration

$$\tilde{C}_i(t_4) = C_i(t_3) + \int_{t_3}^{t_4} \tilde{P}_i(t) dt \quad i = 1, \dots, N_s$$

at  $t_4 = t_3 + \Delta t$ . From the set  $\tilde{C}_i(t_4), i = 1, \dots, N_s$  approximate derivatives  $d/dt \tilde{C}_i(t_4)$  are calculated directly. Then a new polynomial  $\bar{P}_i(t)$  is fitted through  $d/dt C_i(t_1)$ ,  $d/dt C_i(t_2)$ , and  $d/dt C_i(t_3)$ , and  $d/dt \tilde{C}_i(t_4)$ , and interpolated or corrected concentrations

$$\bar{C}_i(t_4) = C_i(t_3) + \int_{t_3}^{t_4} \bar{P}_i(t) dt \quad i = 1, \dots, N_s$$

are computed.

Since the truncation errors for the extrapolation and interpolation formulas are of the same order, the difference  $\bar{C}_i(t_4) - \tilde{C}_i(t_4)$  provides an estimate of the actual truncation error incurred by the integration scheme in the interval from  $t_3$  to  $t_4$ . In fact, it can be shown that, neglecting higher order terms, the true value  $C_i(t_4)$  lies between  $\bar{C}_i(t_4)$  and  $\tilde{C}_i(t_4)$ . Again, see reference (6), chapter 8 for a general discussion of predictor-corrector techniques.

In the algorithm the quantity  $|\bar{C}_i(t_4) - \tilde{C}_i(t_4)|$  is tested to ensure that the integration is sufficiently accurate. A relative tolerance specified through program input, multiplied by  $\sum_{i=1}^{N_s} C_i(0)$ , is taken as an absolute upper limit for

$|\bar{C}_i(t_4) - \tilde{C}_i(t_4)|$ . If for any  $i$  this limit is exceeded, the time interval  $\Delta t$  is halved and the integration retried. The process of testing, cutting the interval, and reapplying the predictor-corrector step is repeated until  $|\bar{C}_i(t_4) - \tilde{C}_i(t_4)|$  is sufficiently small. Then the predicted values  $\tilde{C}_i(t_4), i = 1, \dots, N_s$  are taken as the result of integration, and  $d/dt \tilde{C}_i(t_4), i = 1, \dots, N_s$  as the set of corresponding derivatives. (The predicted rather than the corrected results are accepted to avoid computing the additional set of derivatives.) If, for each  $i, |\bar{C}_i(t_4) - \tilde{C}_i(t_4)|$  is at least one hundred times smaller than the limiting criterion, the

interval  $\Delta t$  will be doubled before the next integration step is begun. The whole process is then repeated over successive intervals, at each step the latest  $\Delta t$  and conditions at the last four points entering the computation, until the end of the time period is reached.

The program was coded in FORTRAN IV for use on the IBM 360 series of computers. Execution times for typical problems on the 360 model 75 computer range between one and two minutes.

**TABLE I**  
**FLUOROCARBON REACTIONS AND RATE CONSTANTS<sup>a</sup>**

R1 $\text{CF}_3\text{I} + \text{M} \rightarrow \text{CF}_3 + \text{I} + \text{M}$	$k_1 = 2.27 \times 10^{30} T^{-4.0} \exp(-57385/\text{RT})$
R2 $\text{C}_2\text{F}_6 + \text{M} \rightarrow \text{CF}_3 + \text{CF}_3 + \text{M}$	$k_2 = 8.40 \times 10^{20} T^{0.5} \exp(-76500/\text{RT})$
R2' $\text{CF}_3 + \text{CF}_3 + \text{M} \rightarrow \text{C}_2\text{F}_6 + \text{M}$	$k_2' = 7.14 \times 10^{17} T^{0.5}$
R3 $\text{CF}_3 + \text{M} \rightarrow \text{CF}_2 + \text{F} + \text{M}$	$k_3 = 1.57 \times 10^{49} T^{-9.04} \exp(-92254/\text{RT})$
R3' $\text{CF}_2 + \text{F} + \text{M} \rightarrow \text{CF}_3 + \text{M}$	$k_3' = 1.49 \times 10^{46} T^{-9.04} \exp(-2287/\text{RT})$
R4 $\text{CF}_4 + \text{M} \rightarrow \text{CF}_3 + \text{F} + \text{M}$	$k_4 = 6.15 \times 10^{34} T^{-4.64} \exp(-122421/\text{RT})$
R4' $\text{CF}_3 + \text{F} + \text{M} \rightarrow \text{CF}_4 + \text{M}$	$k_4' = 9.79 \times 10^{31} T^{-4.64} \exp(-2849/\text{RT})$
R5 $\text{CF}_2 + \text{M} \rightarrow \text{CF} + \text{F} + \text{M}$	$k_5 = 4.20 \times 10^{26} T^{-2.85} \exp(-106000/\text{RT})$
R5' $\text{CF} + \text{F} + \text{M} \rightarrow \text{CF}_2 + \text{M}$	$k_5' = 6.57 \times 10^{26} T^{-2.85}$
M = collision partner (Argon)	

<sup>a</sup>units in calories, cubic centimeter, degree Kelvin, mole, second.

**TABLE 2**  
**EXAMPLES OF CF<sub>3</sub>I AND C<sub>2</sub>F<sub>6</sub> SHOCK TUBE EXPERIMENTS**

Experiment No.	Shock Velocity (mm/ $\mu$ sec)	<sup>a</sup> Shock Temp. (°K)	P <sub>21</sub>	Total Gas Conc'n (mole/cm <sup>3</sup> ) 10 <sup>6</sup>	Apparent Rate Constant (k <sub>3</sub> , cm <sup>3</sup> /mole sec)
1:100 CF <sub>3</sub> I - Argon Data					
1	1.322	1804	3.66	7.94	2.28 x 10 <sup>8</sup>
2	1.382	1928	3.75	8.16	9.08 x 10 <sup>8</sup>
3	1.441	2060	3.82	8.33	4.95 x 10 <sup>9</sup>
4	1.478	2143	3.87	8.45	5.08 x 10 <sup>9</sup>
5	1.510	2214	3.92	8.56	7.73 x 10 <sup>9</sup>
1:100 C <sub>2</sub> F <sub>6</sub> - Argon Data					
6	1.347	1735	4.00	10.8	9.09 x 10 <sup>7</sup>
7	1.395	1850	4.02	8.72	5.69 x 10 <sup>8</sup>
8	1.440	1960	4.03	8.76	1.20 x 10 <sup>9</sup>
9	1.456	2000	4.04	8.76	1.38 x 10 <sup>9</sup>
10	1.533	2205	4.06	8.82	4.88 x 10 <sup>9</sup>

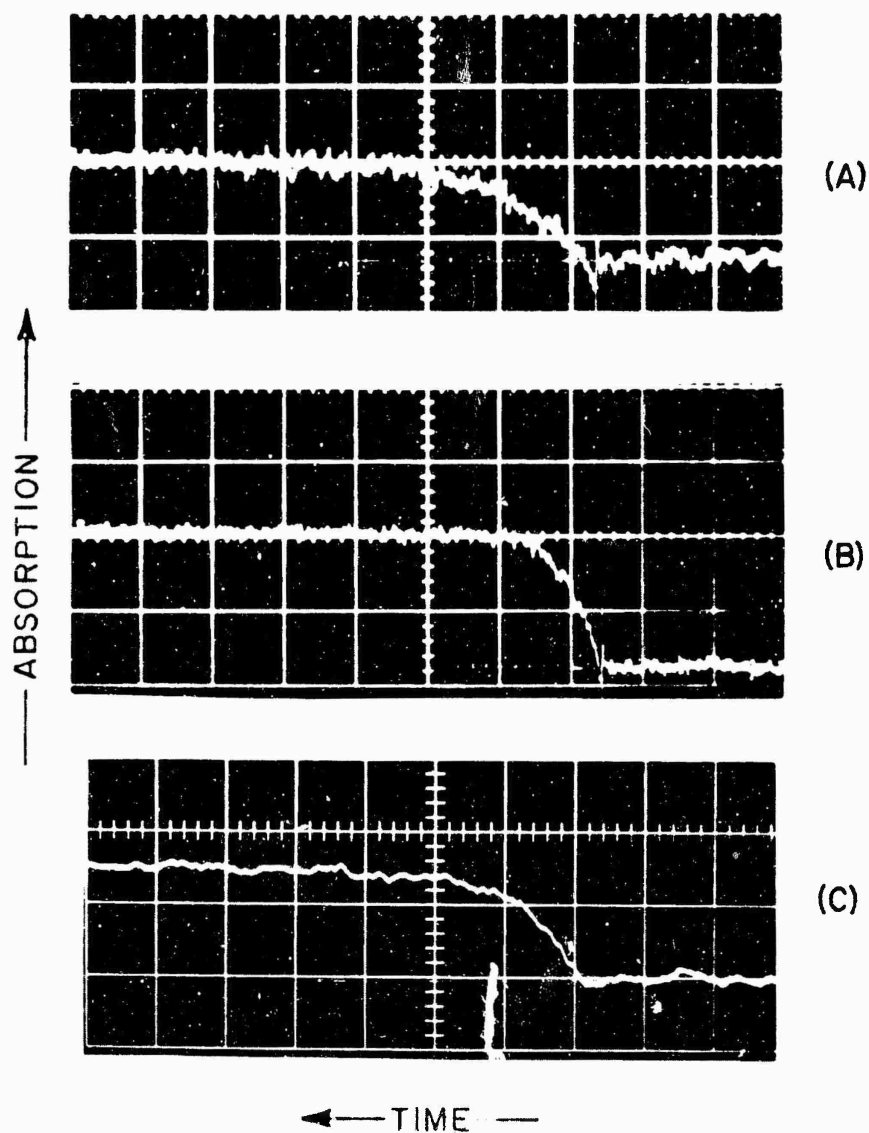
<sup>a</sup>conditions for complete dissociation of CF<sub>3</sub>I and C<sub>2</sub>F<sub>6</sub> at shock front.

TABLE 3

EXAMPLES OF CF<sub>4</sub> SHOCK TUBE EXPERIMENTS

Experiment No.	Shock Velocity (mm/ $\mu$ sec)	<sup>a</sup> Shock Temp. ( $^{\circ}$ K)	$P_{21}$	Gas Conc'n (mole/cm <sup>3</sup> )10 <sup>5</sup>	Apparent Rate Constant ( $k_4$ , cm <sup>3</sup> /mole sec)
1:100 CF <sub>4</sub> - Argon Data					
11	1.559	2410	3.75	1.82	$7.96 \times 10^7$
12	1.623	2590	3.78	1.63	$3.32 \times 10^8$
13	1.643	2640	3.795	1.42	$5.61 \times 10^8$
14	1.670	2720	3.808	0.518	$1.04 \times 10^9$
15	1.715	2850	3.83	0.422	$1.89 \times 10^9$
<sup>a</sup> 16	1.795	2970	4.05	1.09	$4.70 \times 10^9$
2:100 CF <sub>4</sub> - Argon Data					
17	1.507	2260	3.81	1.84	$2.91 \times 10^7$
18	1.544	2350	3.845	1.76	$6.12 \times 10^7$
19	1.591	2460	3.87	1.67	$1.34 \times 10^8$
20	1.623	2550	3.89	1.57	$3.63 \times 10^8$
21	1.66	2650	3.91	1.39	$8.20 \times 10^8$
CF <sub>4</sub> Data					
22	2.015	2145	19.1	0.515	$5.10 \times 10^6$
23	2.075	2248	19.4	0.521	$2.42 \times 10^7$

<sup>a</sup>Rate constant evaluated at 40 percent decomposition.



27-3651

Figure 1 Difluoromethylene ultraviolet light absorption behind shock waves into

(a) 1:100  $\text{CF}_3\text{I-Ar}$ , shock veloc. =  $1.377 \text{ mm}/\mu\text{sec}$ , Temp =  $1984^\circ \text{ K}$

(b) 1:100  $\text{C}_2\text{F}_6\text{-Ar}$ , shock veloc. =  $1.456 \text{ mm}/\mu\text{sec}$ , Temp =  $2107^\circ \text{ K}$

(c) 1:100  $\text{CF}_4\text{-Ar}$ , shock veloc. =  $1.782 \text{ mm}/\mu\text{sec}$ , Temp =  $3050^\circ \text{ K}$

at 4 cm Hg initial pressure. Oscillogram writing =  $20 \mu\text{sec/cm}$

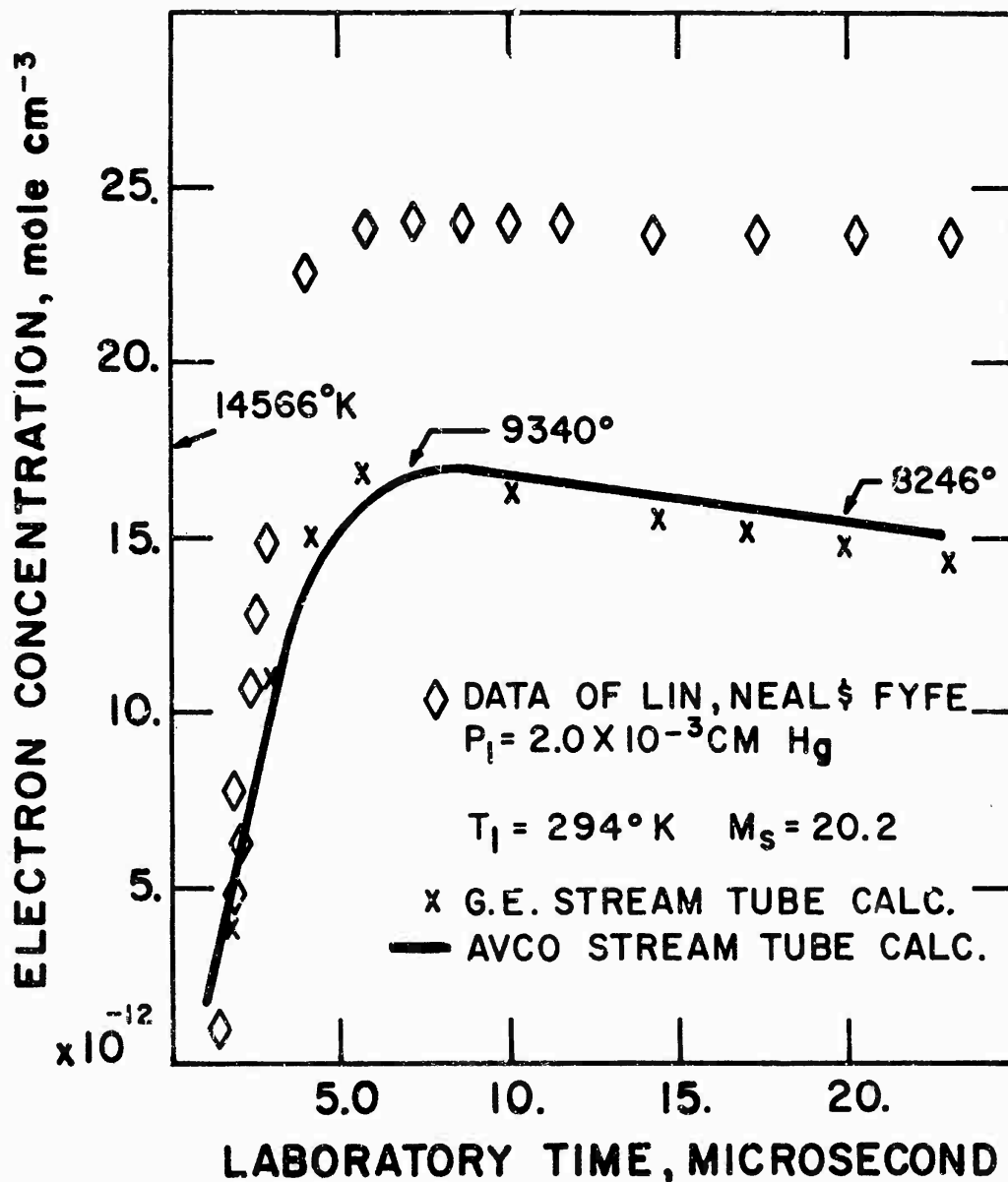
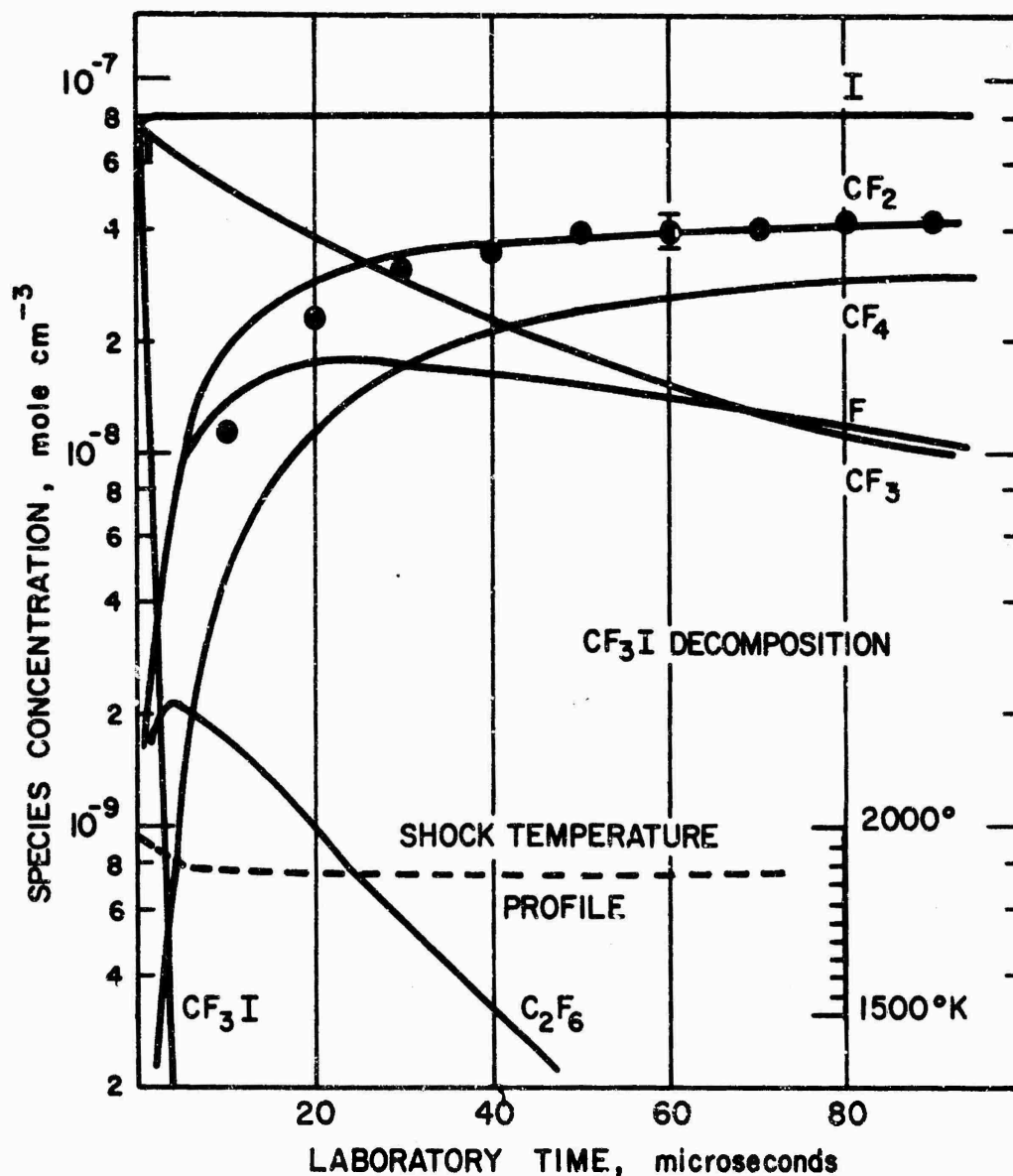
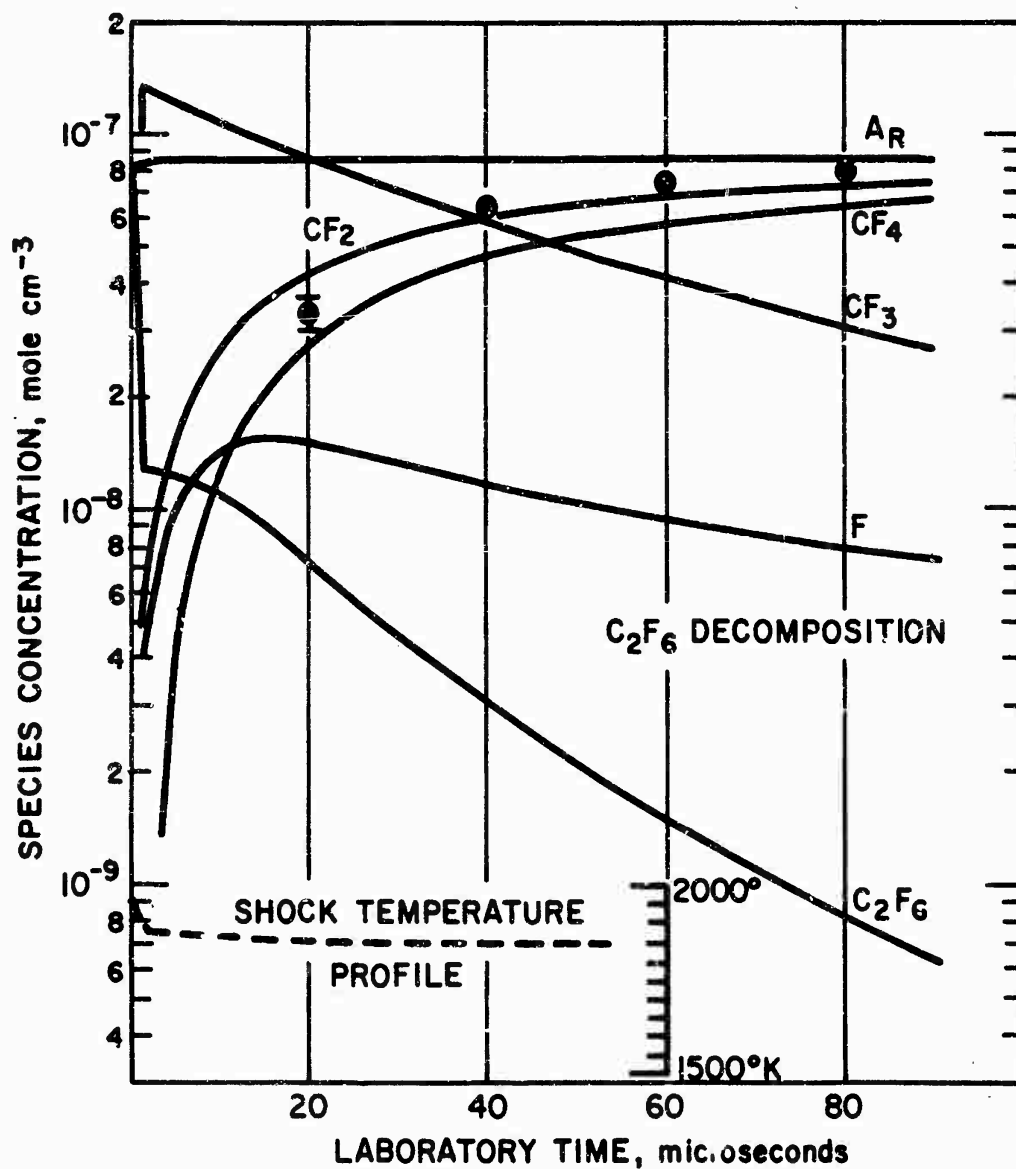


Figure 2 Numerical Check of Nonequilibrium Shock-Tube Program with Airchemistry Stream-Tube Calculation.  
 ◇, ["Rate of Ionization Behind Shock Waves in Air. I. Experimental Results," Avco/Everett Research Rept. 105, (September, 1960)].



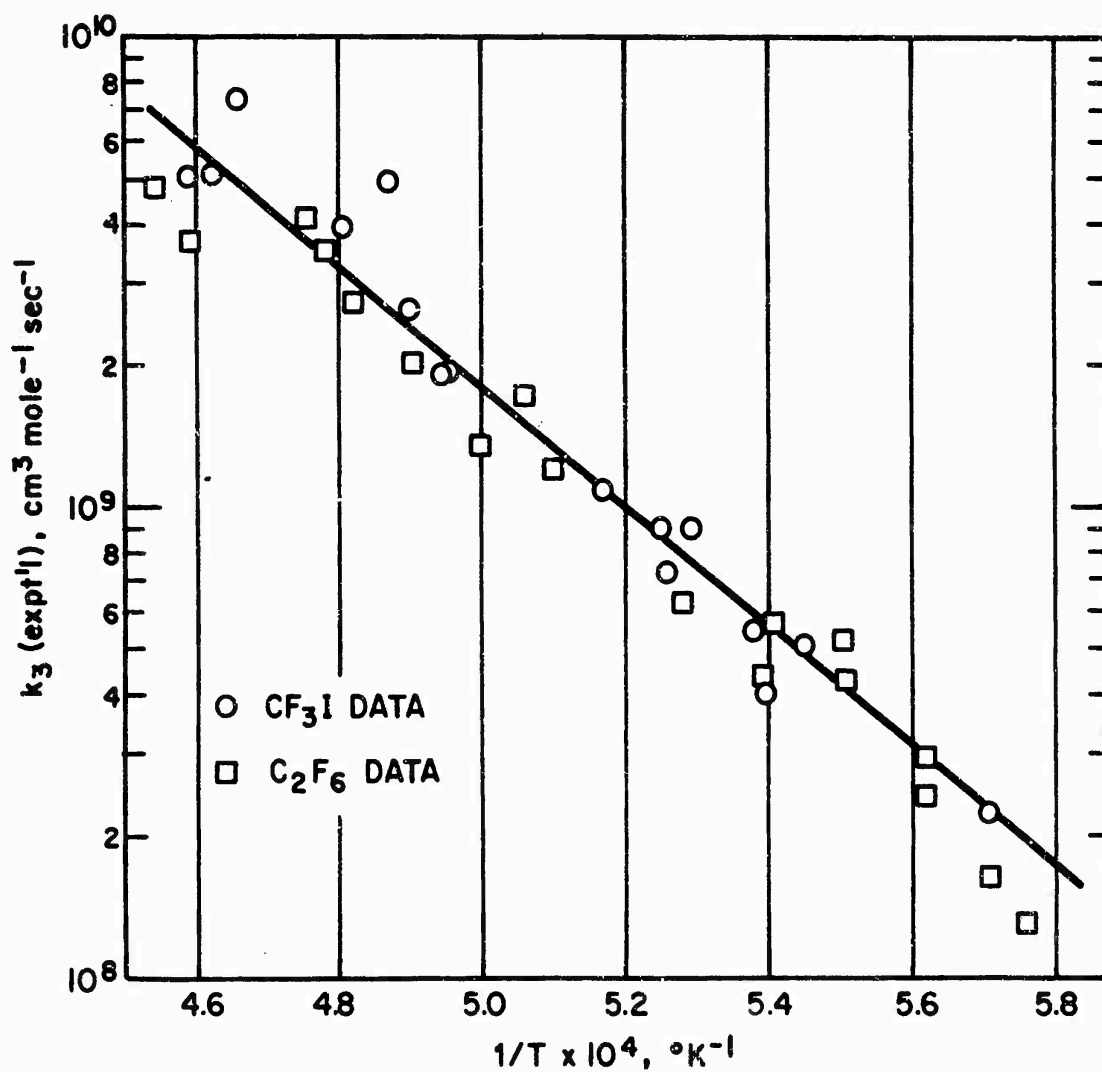
27-3644

Figure 3 Chemical kinetics shock-tube calculation for decomposition of 1:100  $\text{CF}_3\text{I}$ -Argon gas mixture (Experiment No. 2).  $\odot$ , spectroscopic measurements of  $\text{CF}_2$  concentration. Error flag reflects uncertainty in absorption coefficient used in data reduction. Argon curve is (100) I.



27-3645

Figure 4 Chemical kinetics shock-tube calculation for decomposition of 1:100  $C_2F_6$  - Argon gas mixture (Experiment No. 7).  $\odot$ , spectroscopic measurements of  $CF_2$  concentration.

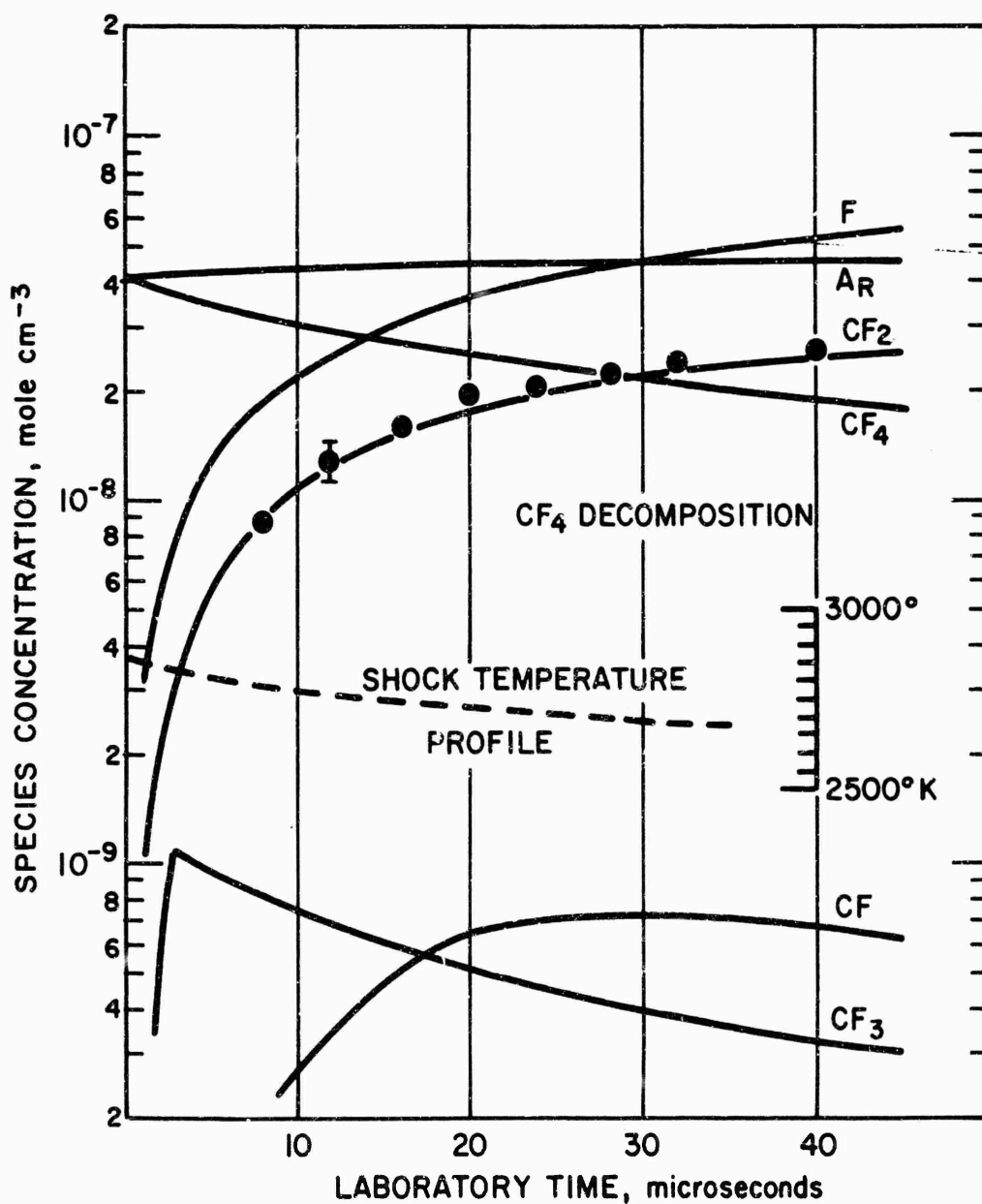


27-3647

Figure 5 Experimental rate constant for CF<sub>3</sub> dissociation.  
Solid line is the function

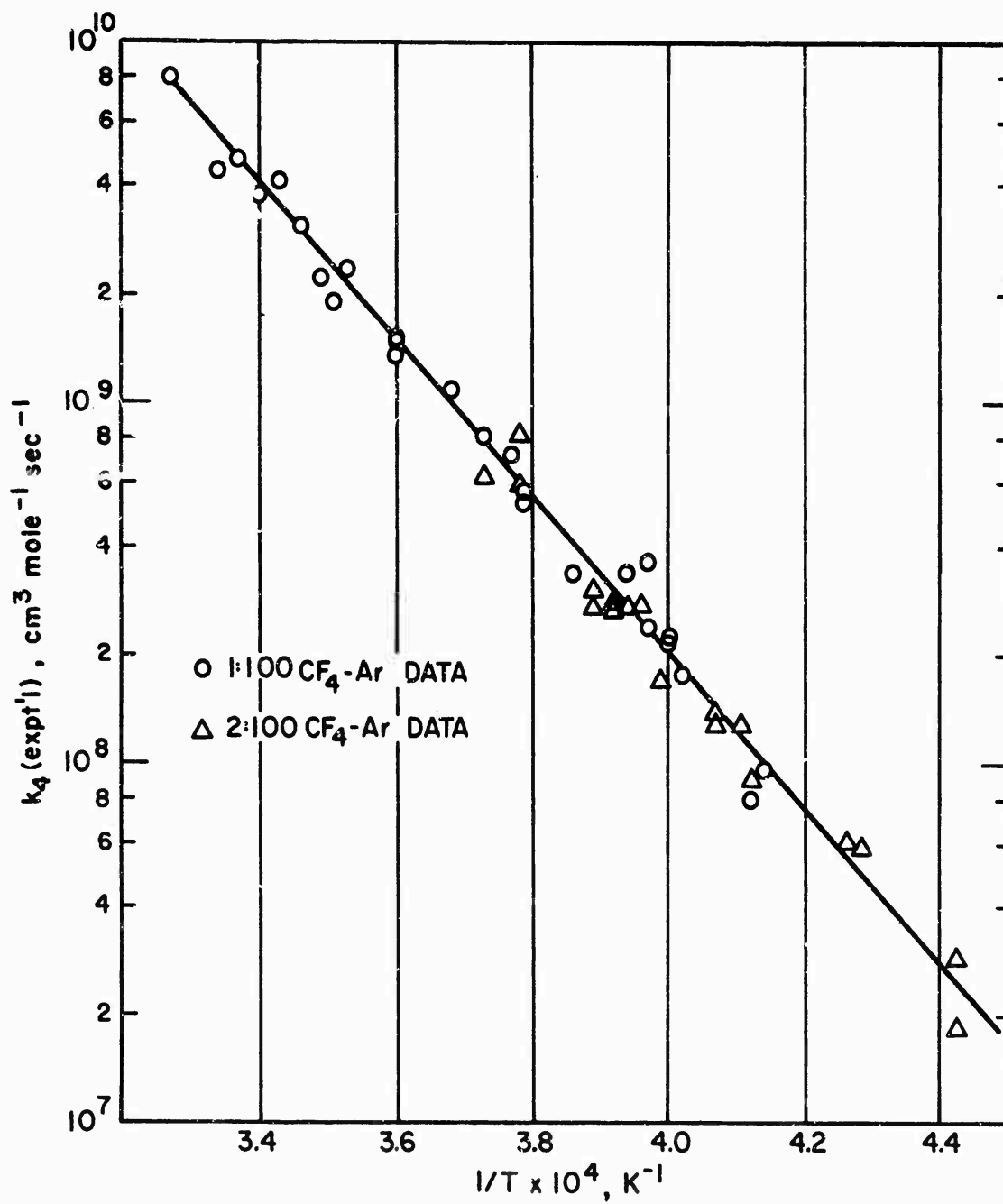
$$1.57 \times 10^{49} T^{-9.04} \exp(-92254/RT)$$

cm<sup>3</sup> mole<sup>-1</sup> sec<sup>-1</sup> from least squares fit of data.



27-3648

Figure 6 Chemical kinetics shock-tube calculation for decomposition of 1:100  $\text{CF}_4$  - Argon gas mixture (Experiment No. 15).  $\odot$ , spectroscopic measurements of  $\text{CF}_2$  concentration.

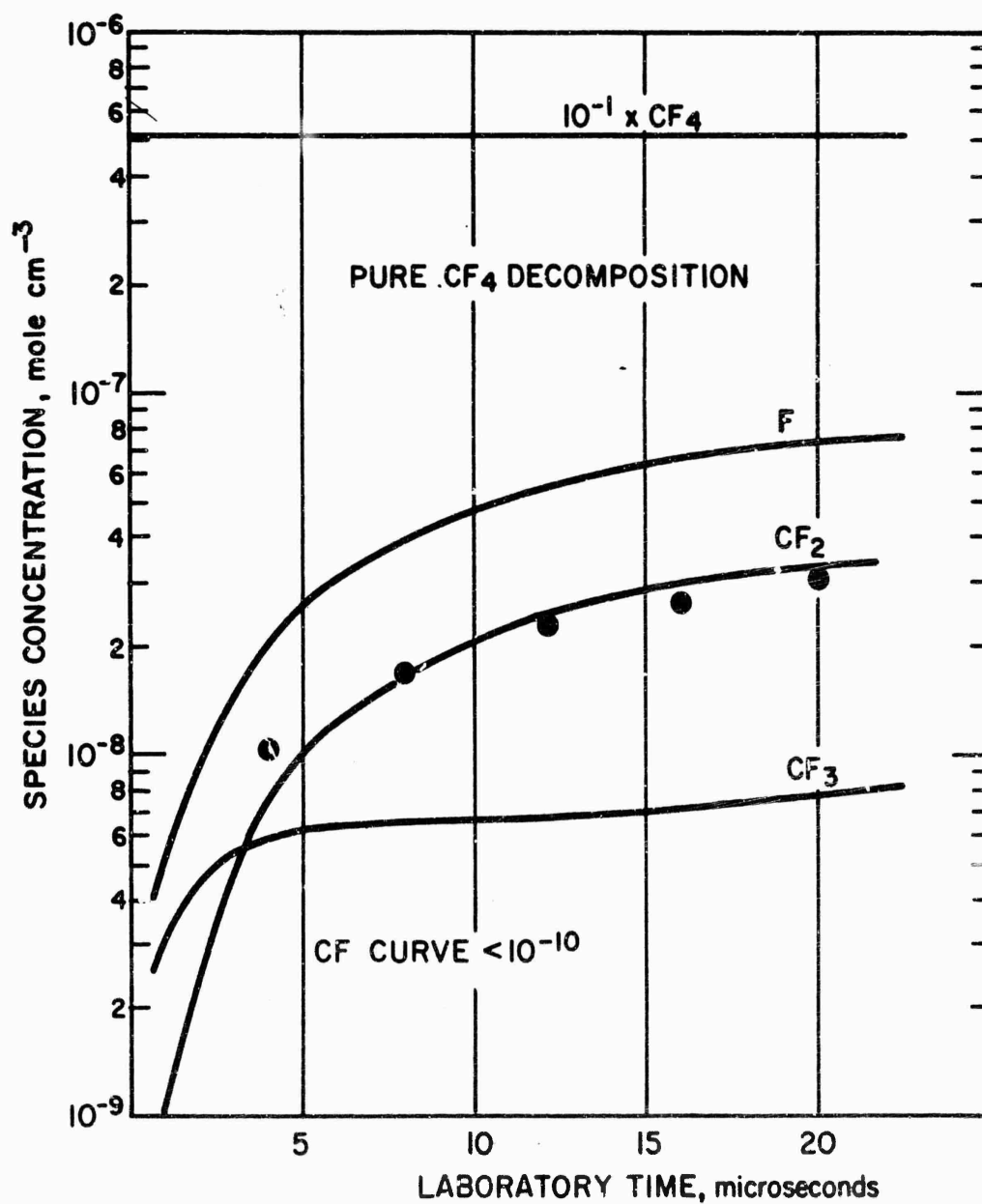


27-3649

Figure 7 Experimental rate constant for CF<sub>4</sub> dissociation.  
Solid line is the function

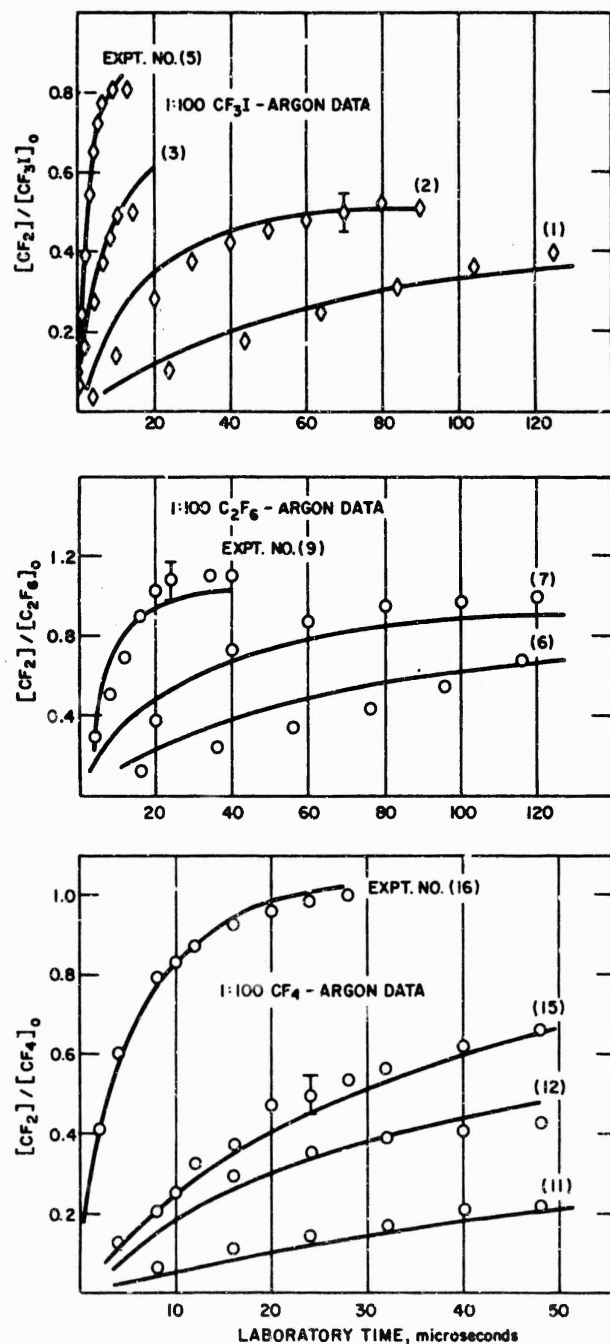
$$6.15 \times 10^{34} T^{-4.64} \exp(-122421/RT)$$

cm<sup>3</sup> mole<sup>-1</sup> sec<sup>-1</sup> from least squares fit of data.



27-3646

Figure 8 Chemical kinetics shock-tube calculation for decomposition of pure  $\text{CF}_4$  (Experiment No. 22).  $\odot$ , spectroscopic measurements of  $\text{CF}_2$  concentration in reaction mixture.



27-3650

Figure 9 Kinetic profiles of  $\text{CF}_2$  in shock heated fluorocarbon-argon gas mixtures. Curves are calculated by the nonequilibrium shock-tube program. Symbols are spectroscopic measurements. Subscript 0 denotes reactant concentration initially behind shock wave.